

WYVERN



42nd Annual VFS Student Design Competition

Pioneering Hydrogen-Electric VTOL

Sponsored by: **AIRBUS**



UNIVERSITY OF
MARYLAND

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Academic Course:
ENAE634: Helicopter Design (3 credits)



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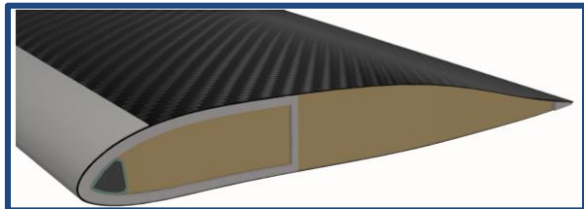
To the Vertical Flight Society:

The members of the University of Maryland Graduate Student Design Team hereby grant VFS full permission to distribute the enclosed Executive Summary and Final Proposal for the 42nd Annual Design Competition as they see fit.

Thank you,
The UMD Graduate Design Team



Wyvern: Key Features



**Lightweight
Composite
Blades**



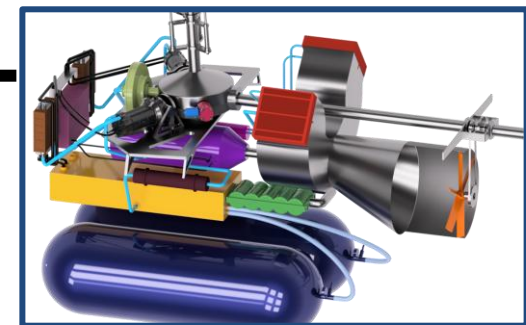
**Simple and Robust
Articulated Hub**



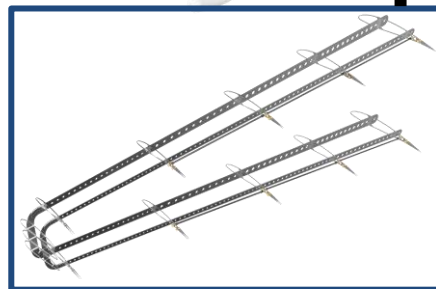
**Ultra Light Weight
Drivetrain**



Modern Glass Cockpit



**Efficient PEMFC
Powerplant**



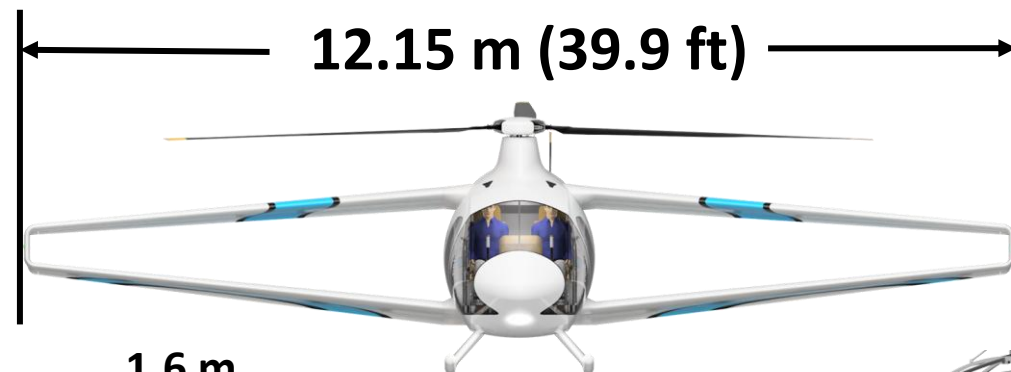
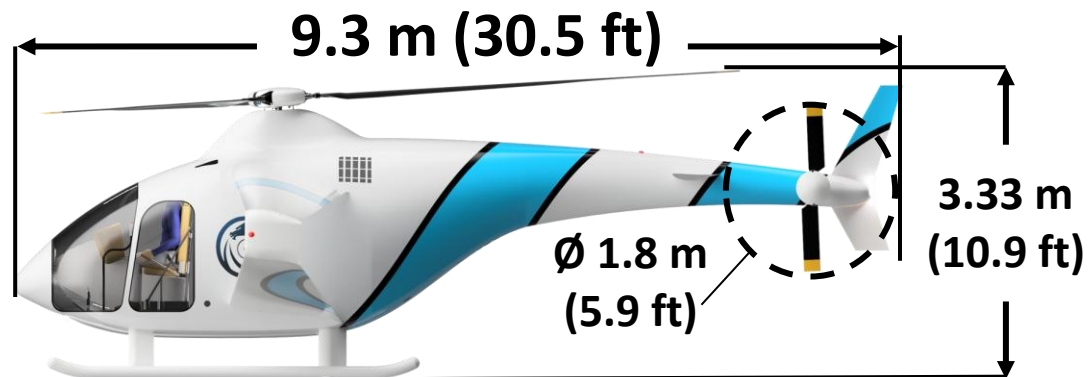
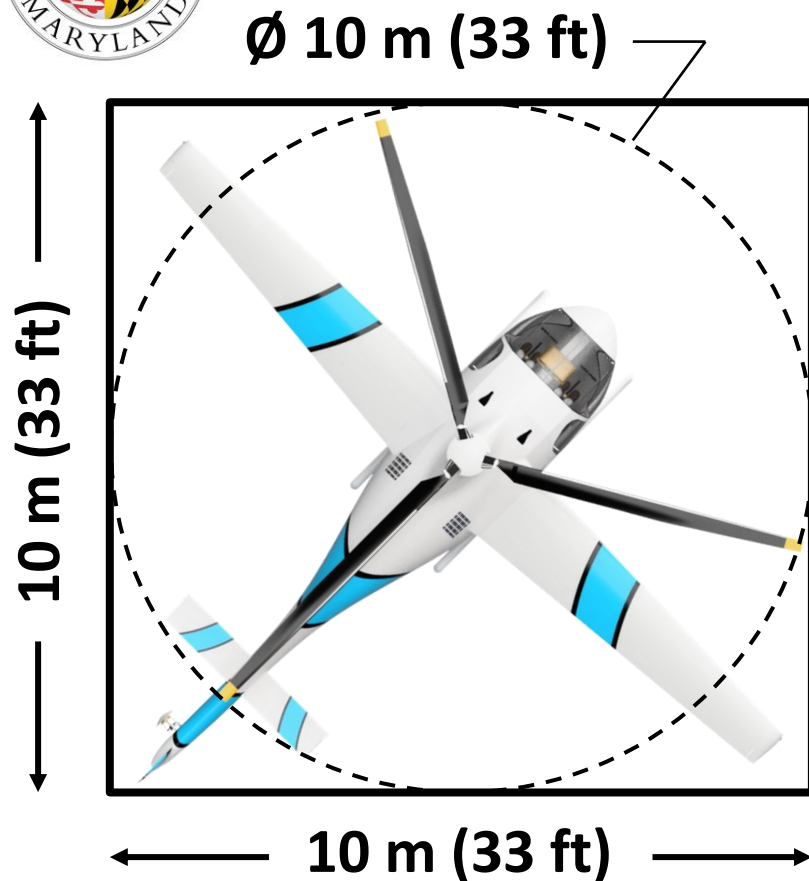
**Aerodynamically
Efficient Box Wing**



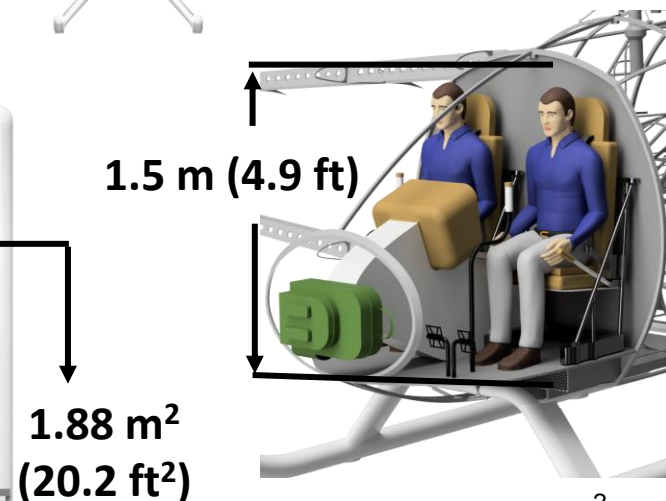
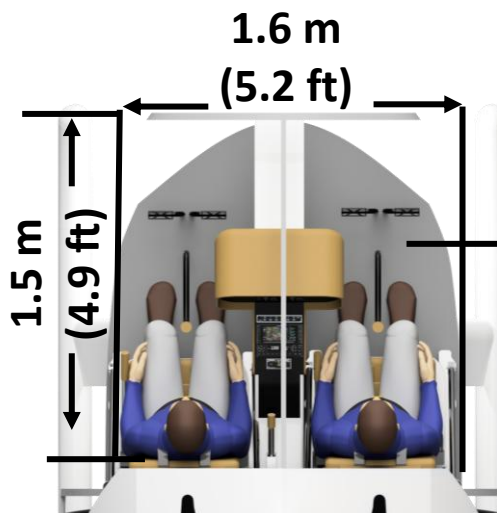
**Compact Floiler
Mechanism**



Vehicle Overview



Characteristics	Value
GTOW	1648 kg (3633 lb)
Installed Power	277 kW (371 hp)
Hydrogen Mass	24.8 kg (55 lb)
Loiter Endurance	4.5 hours @ 57 knots
Payload	185 kg (408 lb)





WYVERN: A Flame-Free Hydrogen Hybrid Helicopter



Like the majestic, agile, and flame-free mythical dragon that lends its name, **Wyvern** rises as the first manned hydrogen-powered all-electric fuel cell compound rotorcraft.

With a **1648 kg GTOW**, **277 kW installed power**, **5 m radius rotor** and a highly efficient **box-wing**, **Wyvern** achieves **over four hours of zero-emission flight time** with a modest **25 kg of gaseous H₂** and a large **185 kg dual-occupant payload**.



A **composite skin**, **lightweight low-vibration articulated rotor hub**, **super-critical driveshaft**, **high-endurance gearing**, **weight-efficient wing structure**, **custom-designed PEM fuel cell stack**, and **hybrid battery** compress the **Wyvern** empty weight fraction to **47%**. The all-electric drive train achieves over **35% tank-to-shaft efficiency** while a **210 kW PEM fuel-cell stack**, boosted by a **high-C-rate Li-ion battery** with silicone anode, feeds **twin 180 kW high power-density motors**.

Crash-worthy **700 bar Type-IV hydrogen tanks** sit behind fire barriers, **Nomex/Kevlar honeycomb mesh** protects the cockpit and tanks, **dual-loop cooling** keeps the PEM stack at 80°C, and simplex hydraulic actuated flight-control channels pair with predictive health-monitoring across all critical systems.

Combined with **fail-safe autorotation capability** and compliance with **EASA CS-27 guidance at TRL ≥ 6**, these features provide pilots with the endurance of a fixed-wing platform and the agility of a helicopter, along with additional payload capacity for sensors or supplies, which lets **Wyvern** patrol sensitive habitats in near silence.

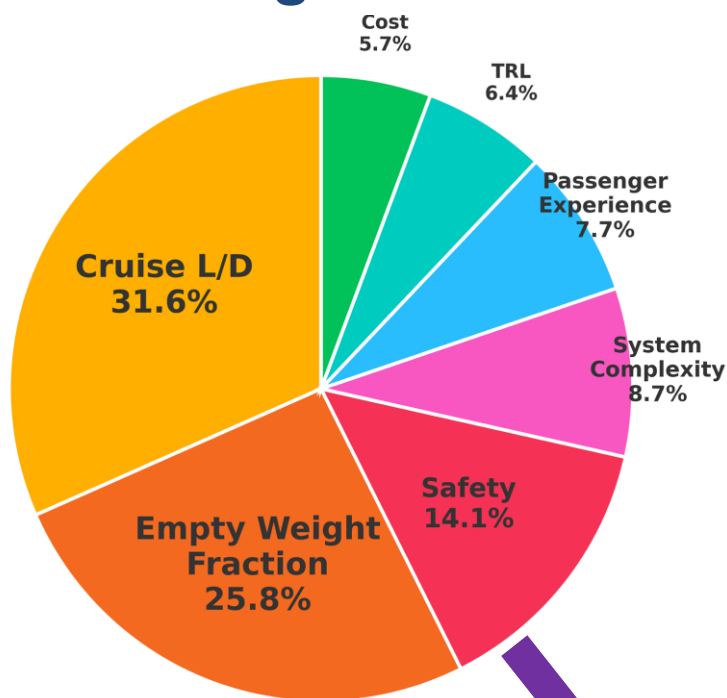
Not Feasible



Feasible With Significant Innovation



Design Drivers



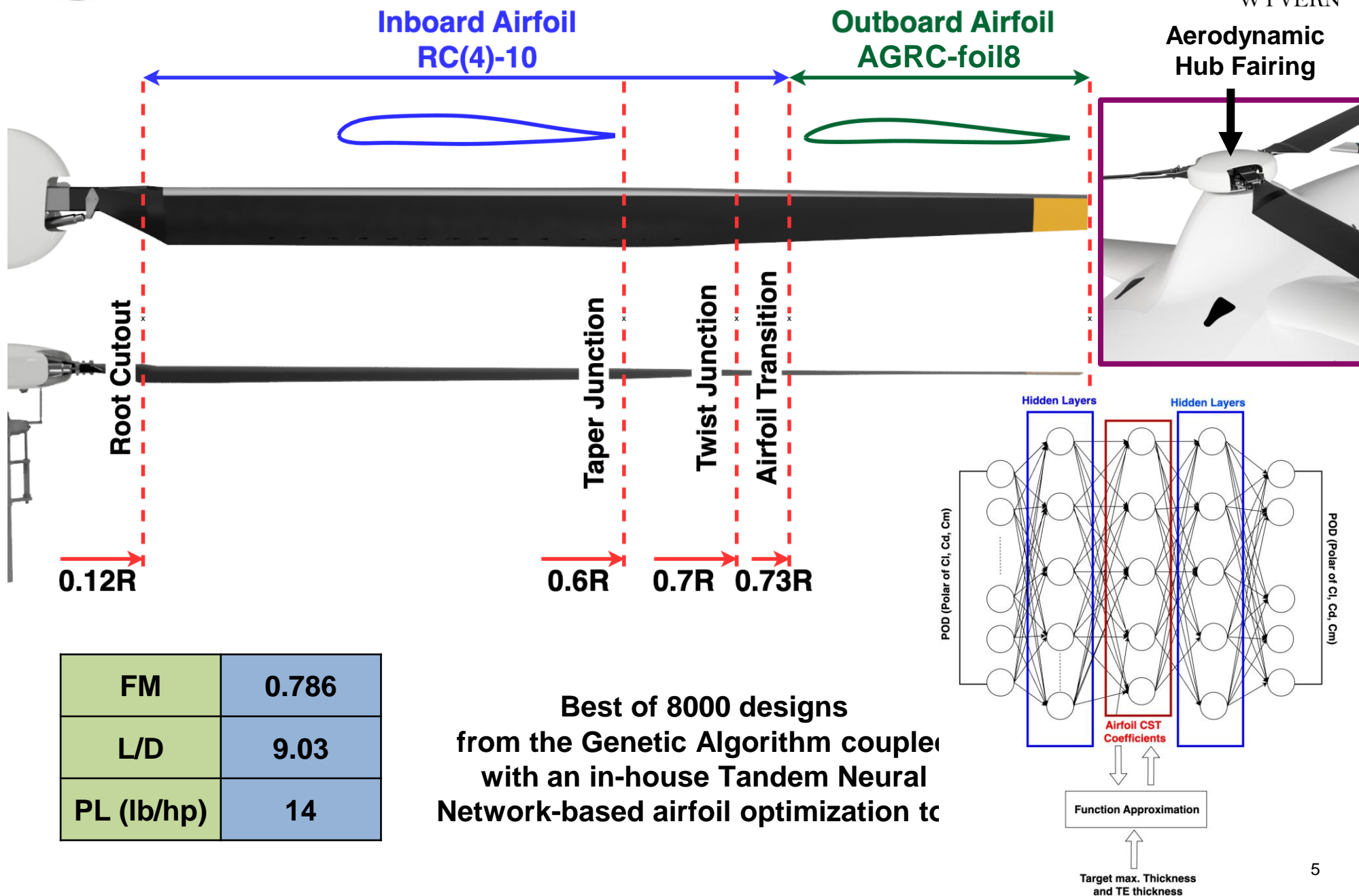
UMD's Innovative Solution

- Rotor RPM slowed by 33% in Loiter
- Box wing with high L/D of 16.5
- Custom *floilers*
- Custom-designed PEM fuel cell stack



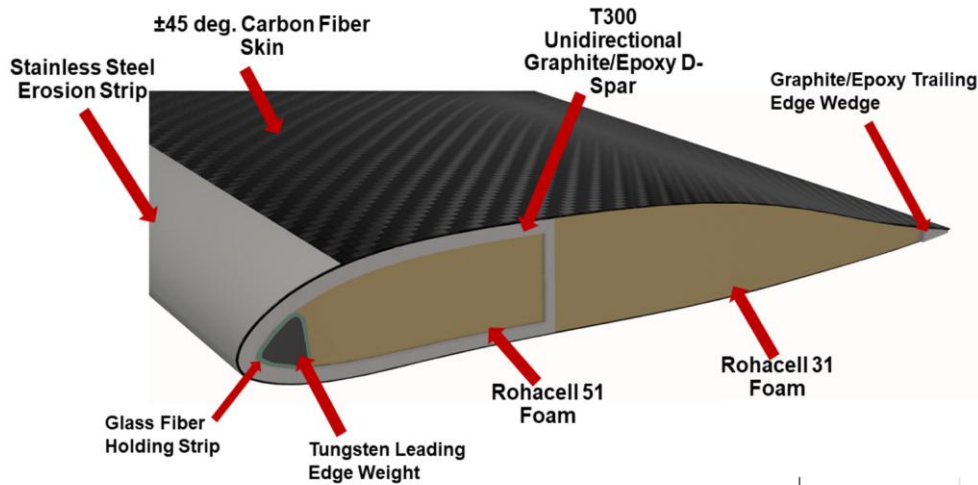


Aerodynamically Optimized Main Rotor



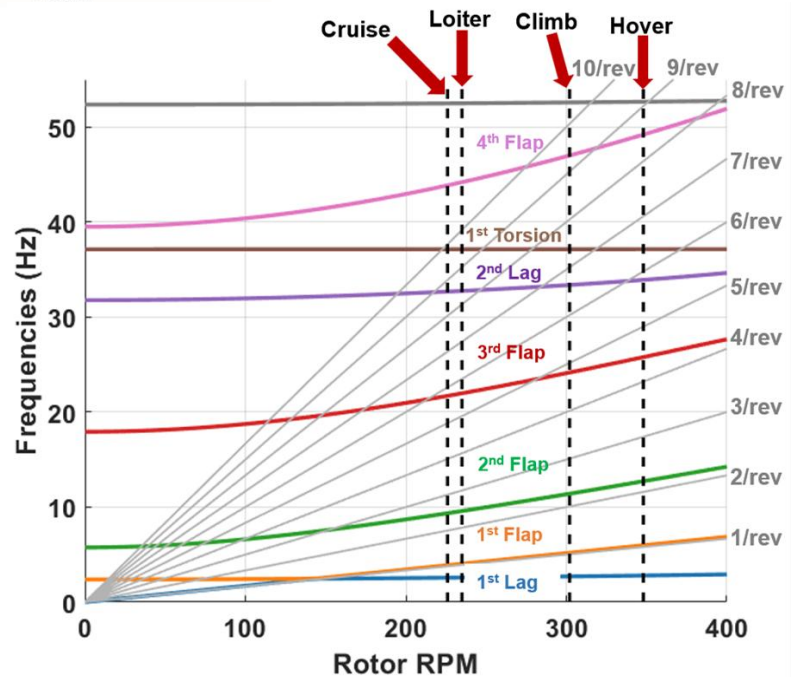
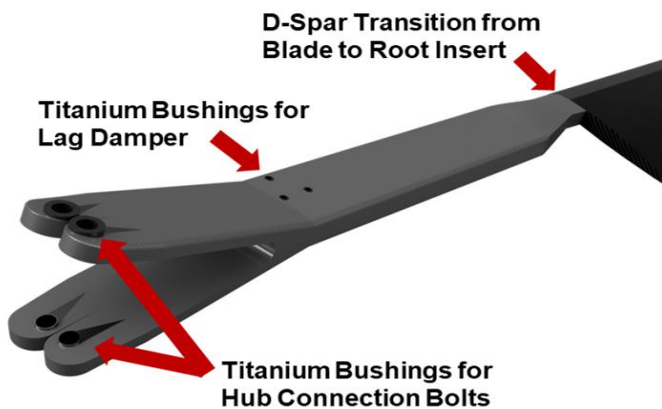


Variable RPM Ground Resonance-Free Rotor

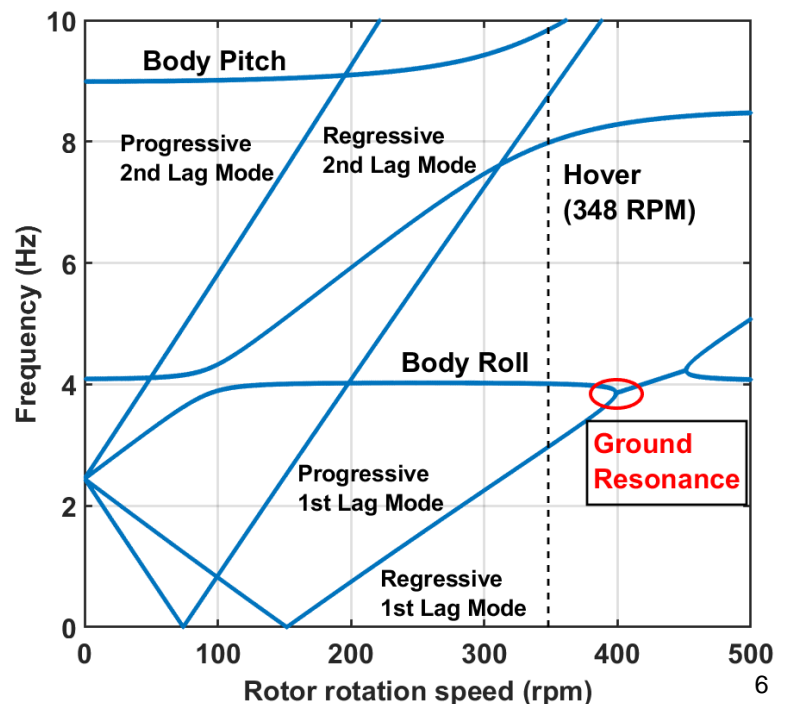


Antinodal tuning mass clears the 2nd Flap from resonance

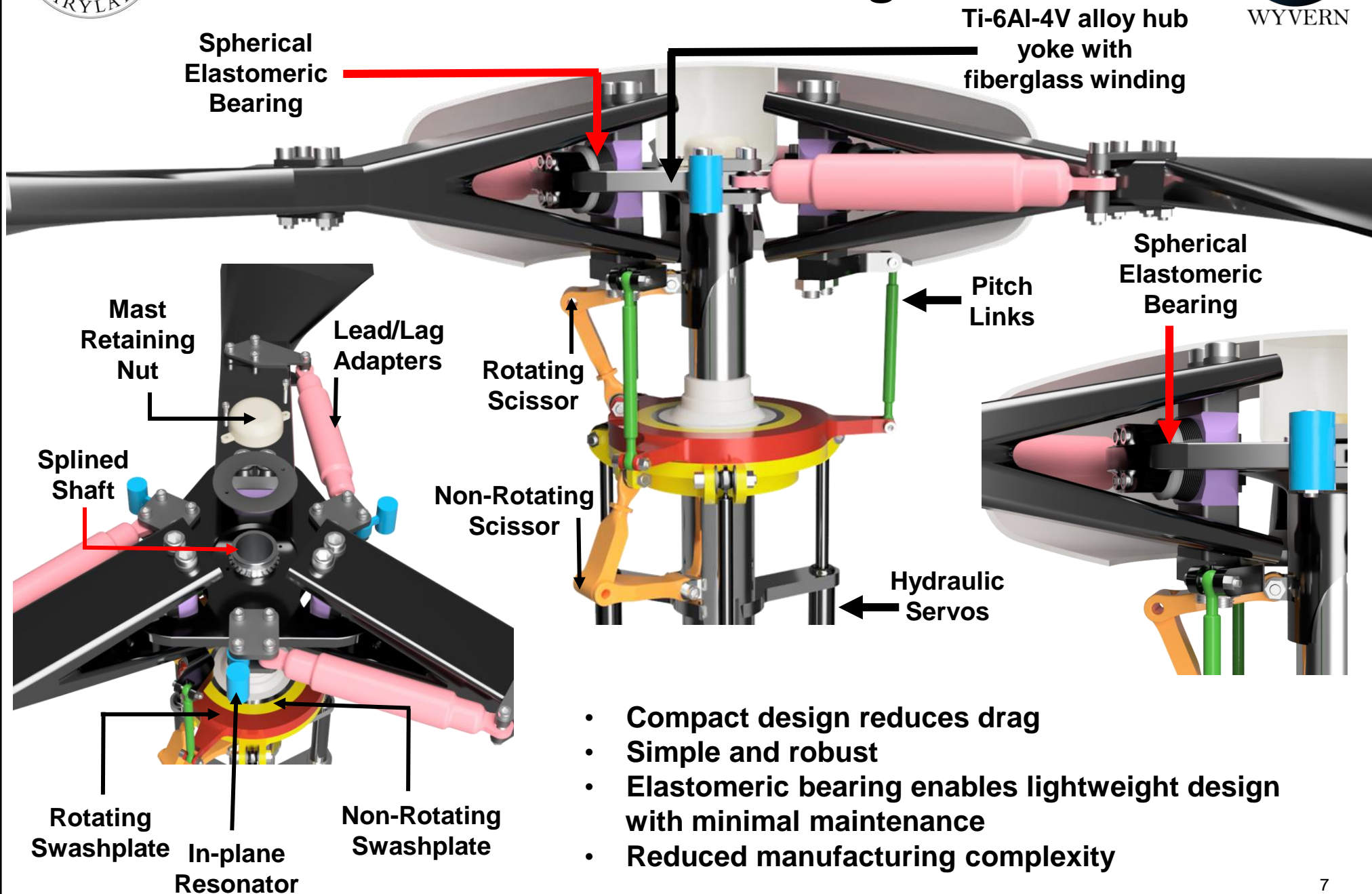
Forked cuff to absorb CF



Omega (RPM)	1 st Lag (/rev)	1 st Flap (/rev)	1 st Torsion
348 (Hover)	0.49	1.03	6.40
302 (Climb)	0.54	1.03	7.38
235 (Loiter)	0.66	1.03	9.48
226 (Cruise)	0.68	1.03	9.86



Articulated Hub with Low-Maintenance Elastomeric Hinges

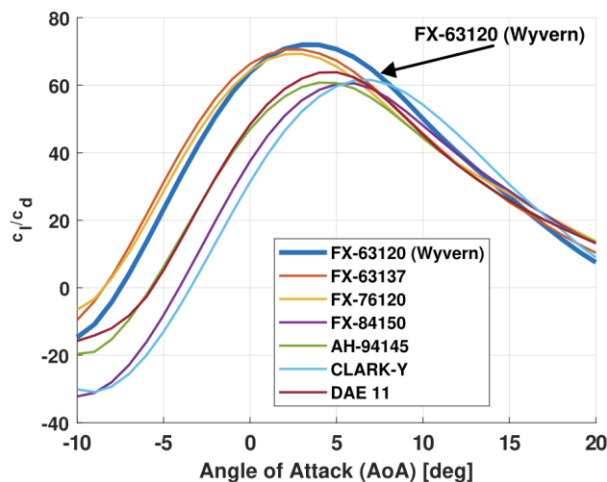
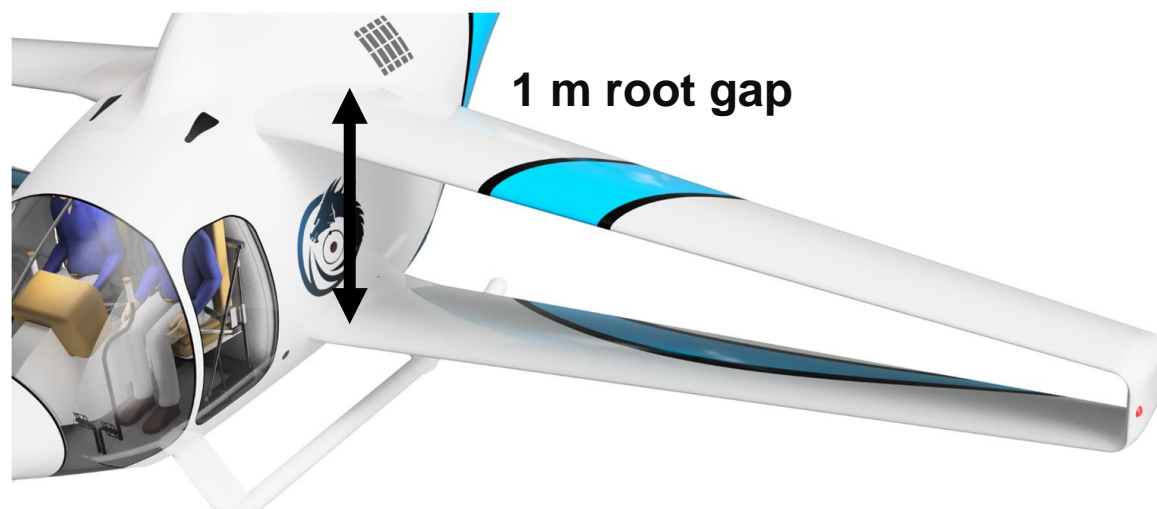
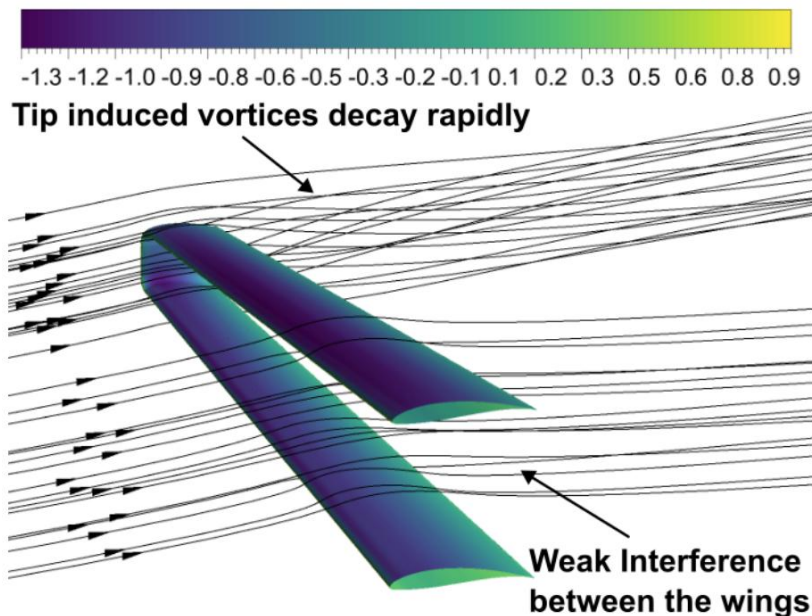


- Compact design reduces drag
- Simple and robust
- Elastomeric bearing enables lightweight design with minimal maintenance
- Reduced manufacturing complexity

Tapered, Tuned Incidence Box Wing

Innovative box-wing offloads 55% of rotor lift during loiter

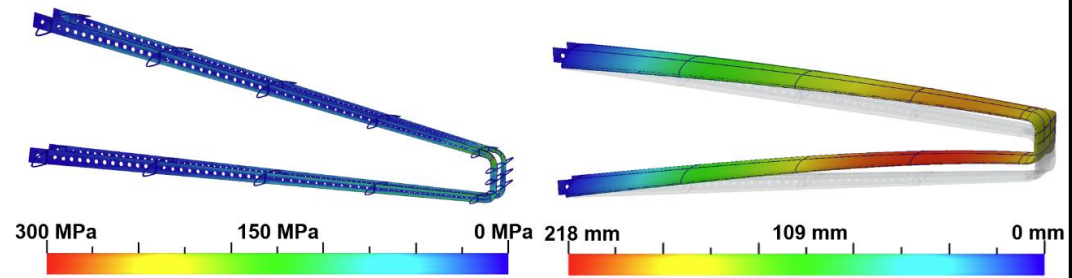
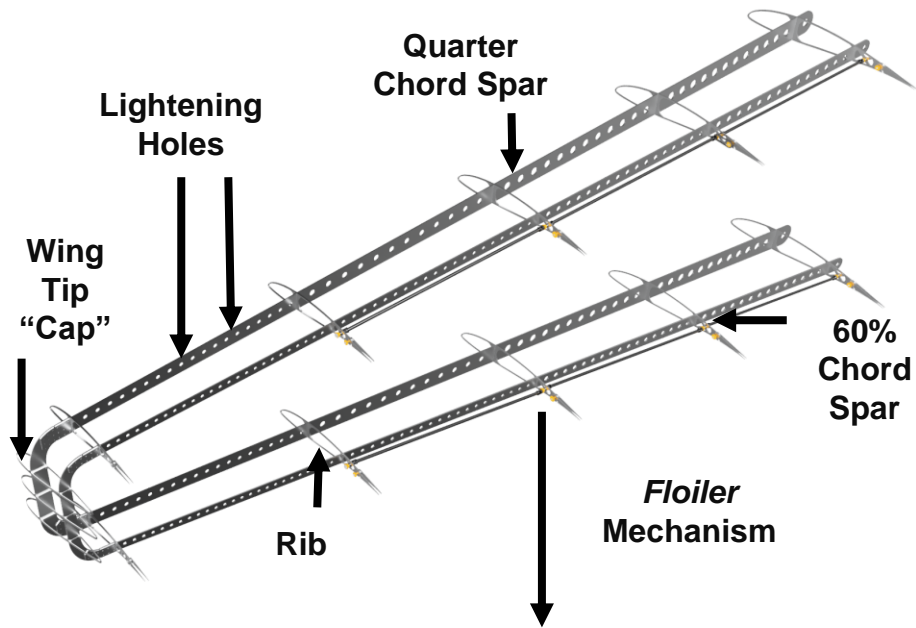
Cp Contours



- 2:1 Taper promotes efficient lift distribution
- Tuned wing incidences for equal wing lift share
- Gap optimized to reduce aerodynamic interference
- Induced drag reduced by 18% compared to a planar wing
- Zero stagger reduces download
- Special airfoil FX 63-120 minimizes load and stabilizer

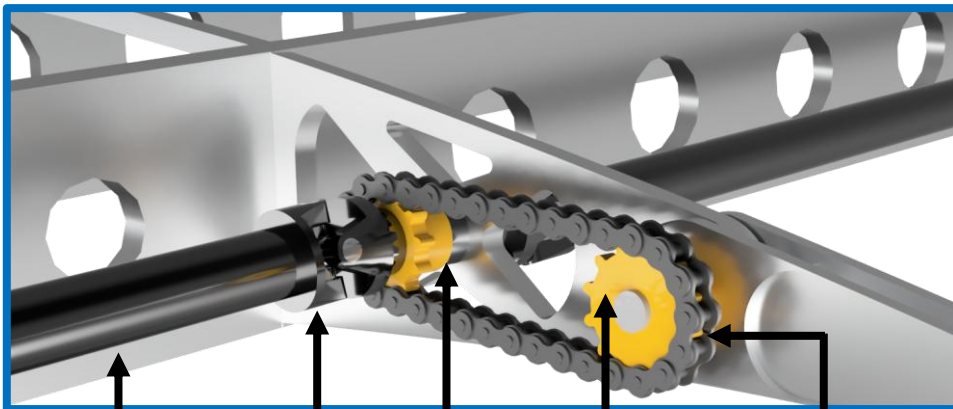


Dual Spar Box Wing Structure With *Floilers*



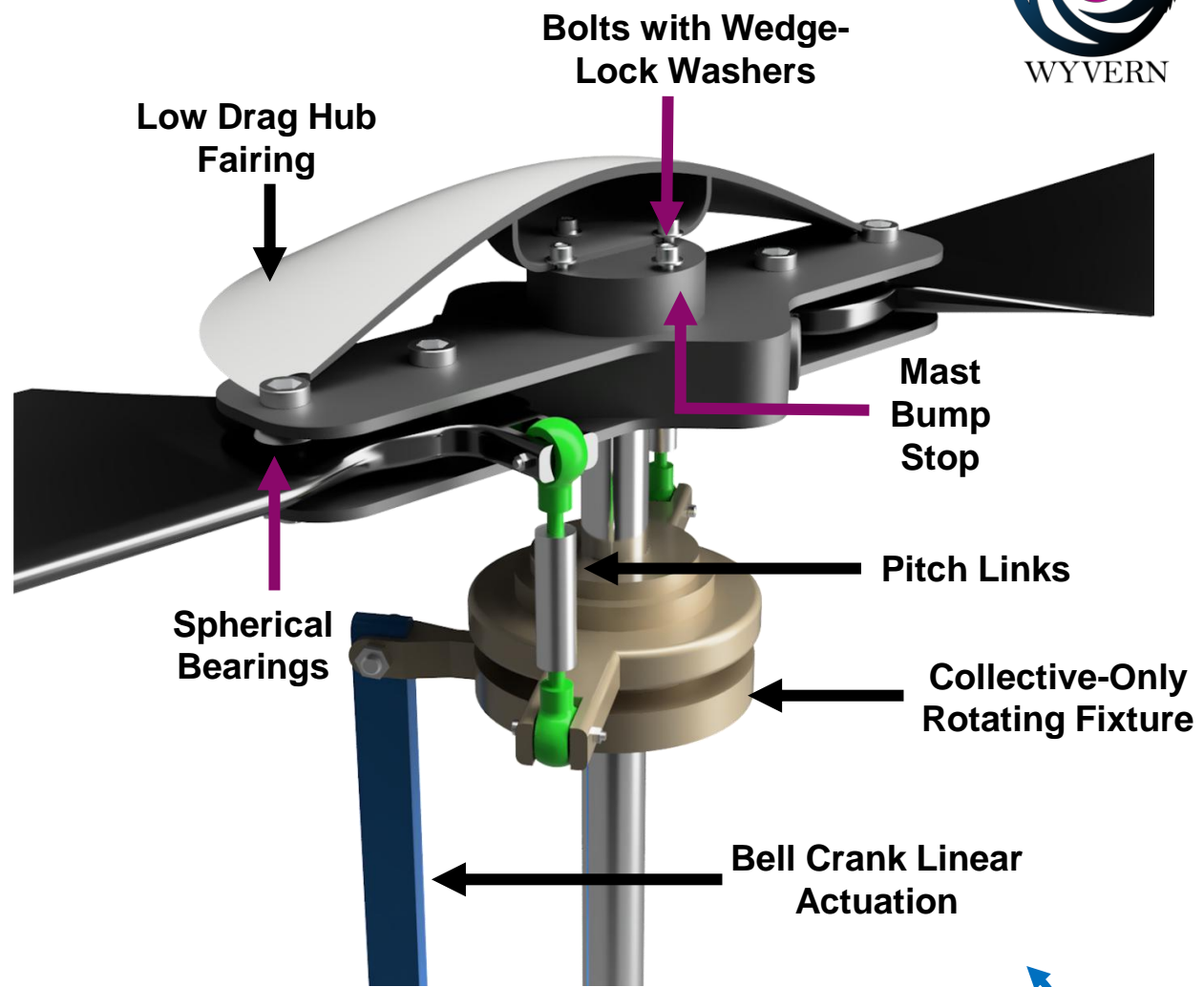
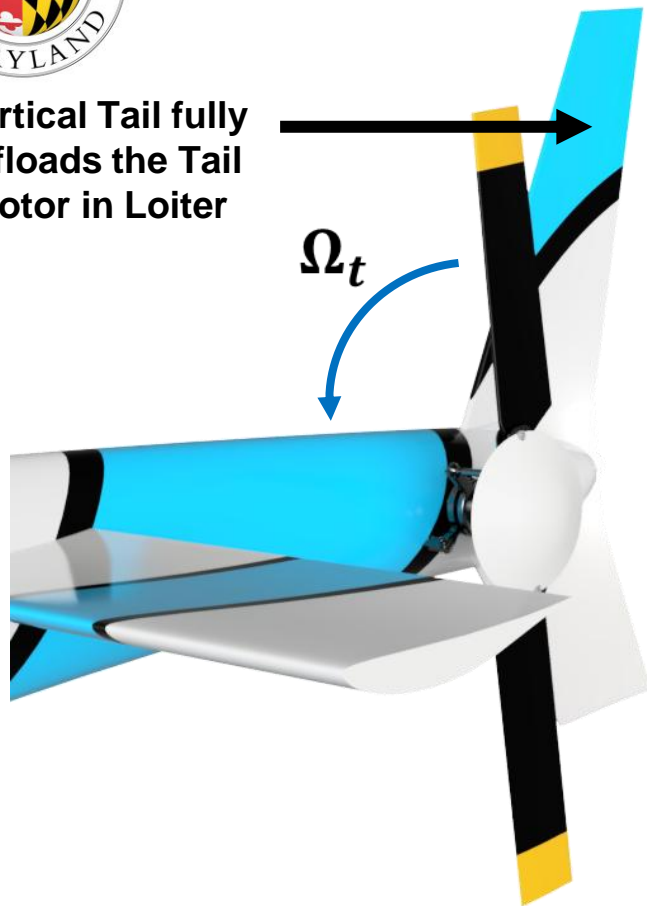
Wyvern wing attached to fuselage through hinges.

Wyvern box wing features a *floiler* that reduces lift during steep descent (deflection up) and enhances hover performance (deflection down)

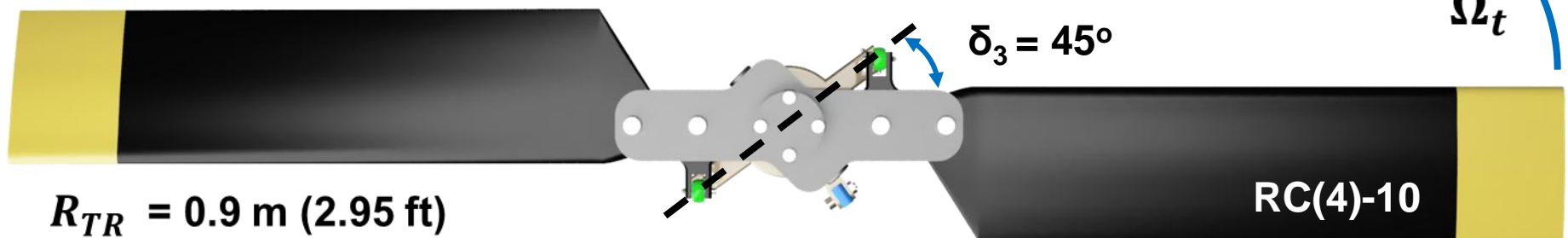


Tail Rotor and Empennage

Vertical Tail fully offloads the Tail Rotor in Loiter



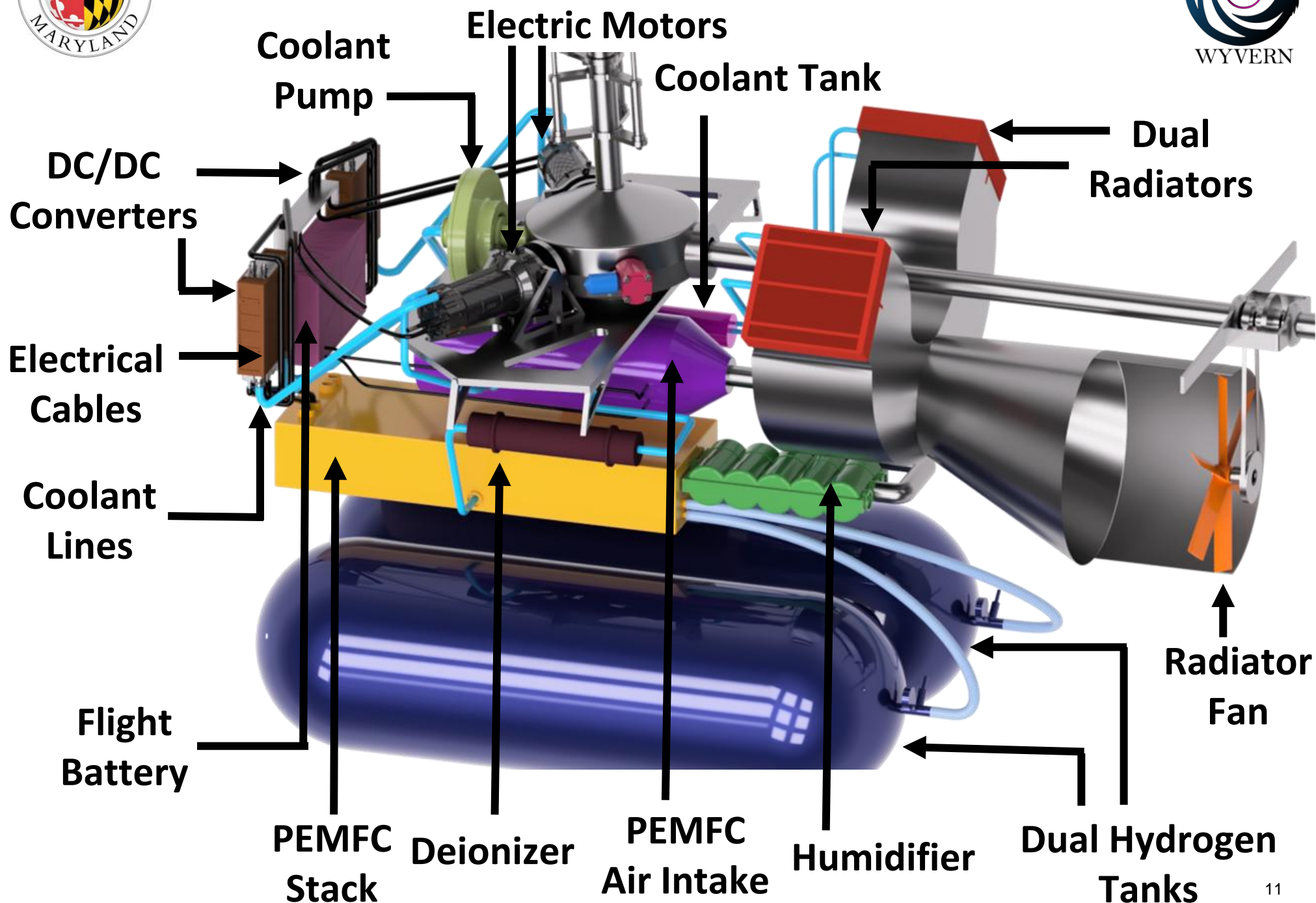
Aerodynamic optimization of the Tail Rotor ensured 0.79 FM



$$R_{TR} = 0.9 \text{ m (2.95 ft)}$$



Integrated PEMFC Powerplant Architecture



Detailed Stack and BOP Design

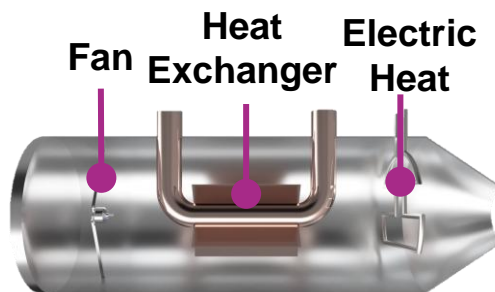
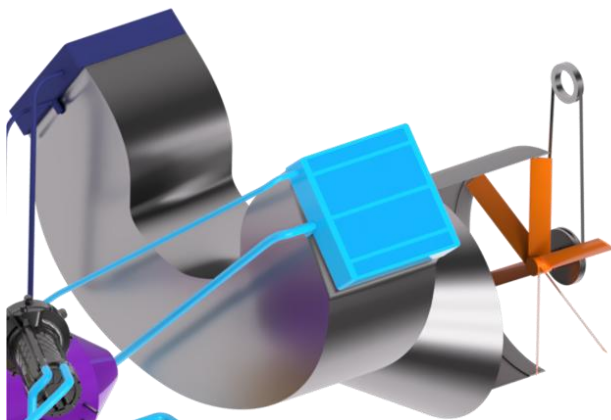
Reliable, weight-saving Humidifier

- Hollow fiber technology passively humidifies
- Eliminates need for water tank and spray nozzles



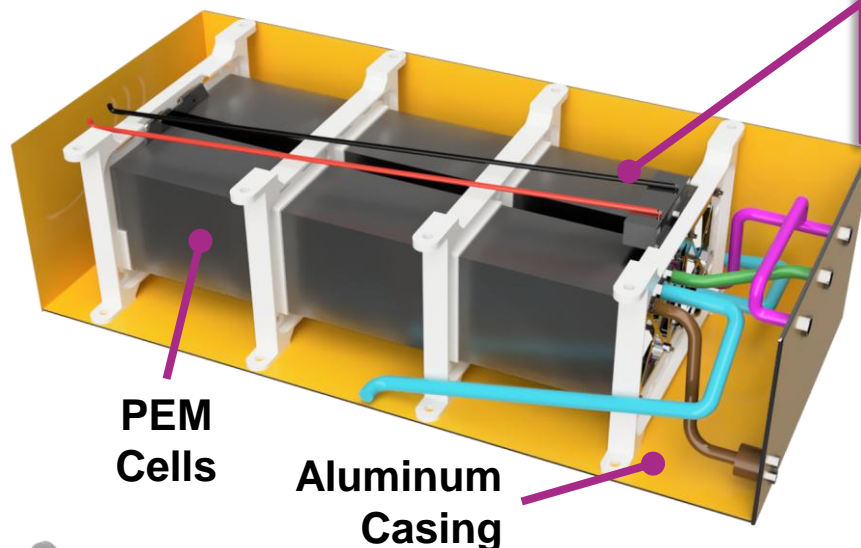
Integrated Cooling System

- Dual radiators remove up to 300 kW of heat
- Simple control via mechanical thermostats



Low Power Air Intake

- Provides up to 0.32 kg/s of warm air

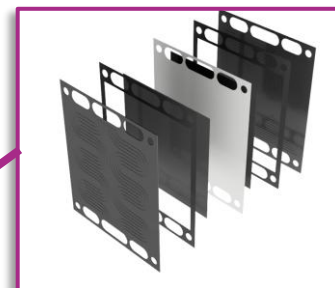


PEM Cells

Aluminum Casing

Lightweight PEM Cell

- 112 grams
- Nafion™ PEM
- Graphite Bi-polar plates
- 327 cm² Active Area



Rugged PEMFC

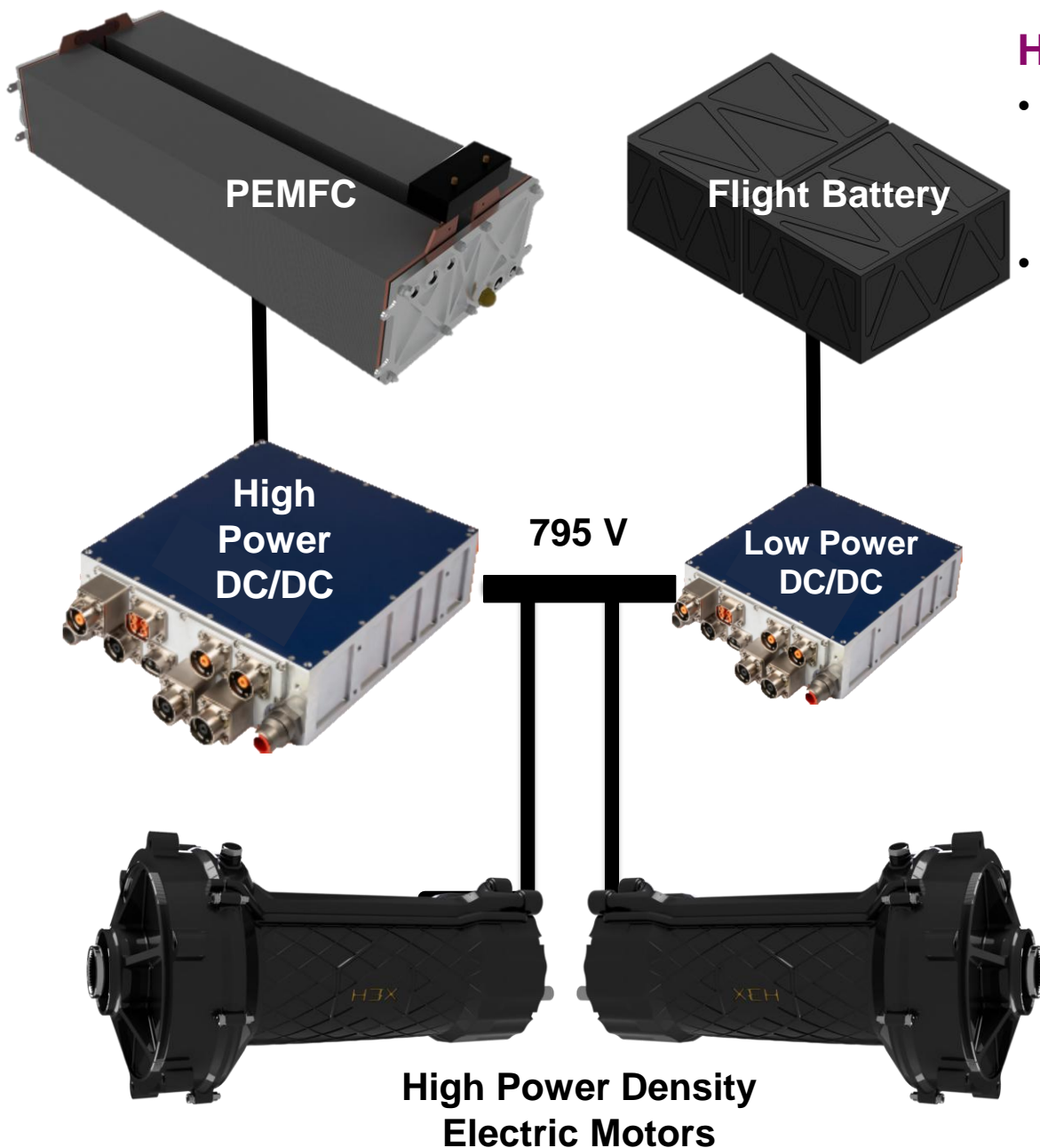
- MCP of 210 kW
- Mass of 127 kg
- 824 Cells
- 1 atm stack pressure eliminates need for air compressor
- Vibration isolating mounts

Safe and Compact Hydrogen Storage

- 2x Type IV, 700 bar tanks with 24.8 kg of storage
- Tanks equipped with pressure release valve, hydrogen detector, and energy absorbing honeycomb



Endurance Boosting Plug-In Hybrid Architecture

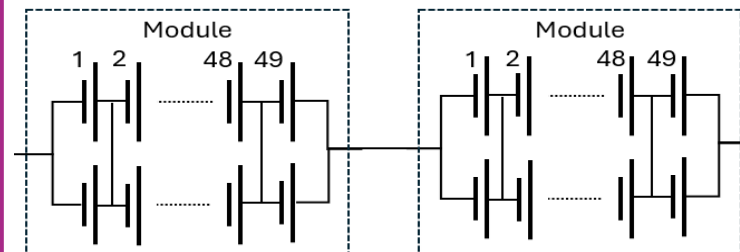


Hover Enhancing Parallel Propulsion

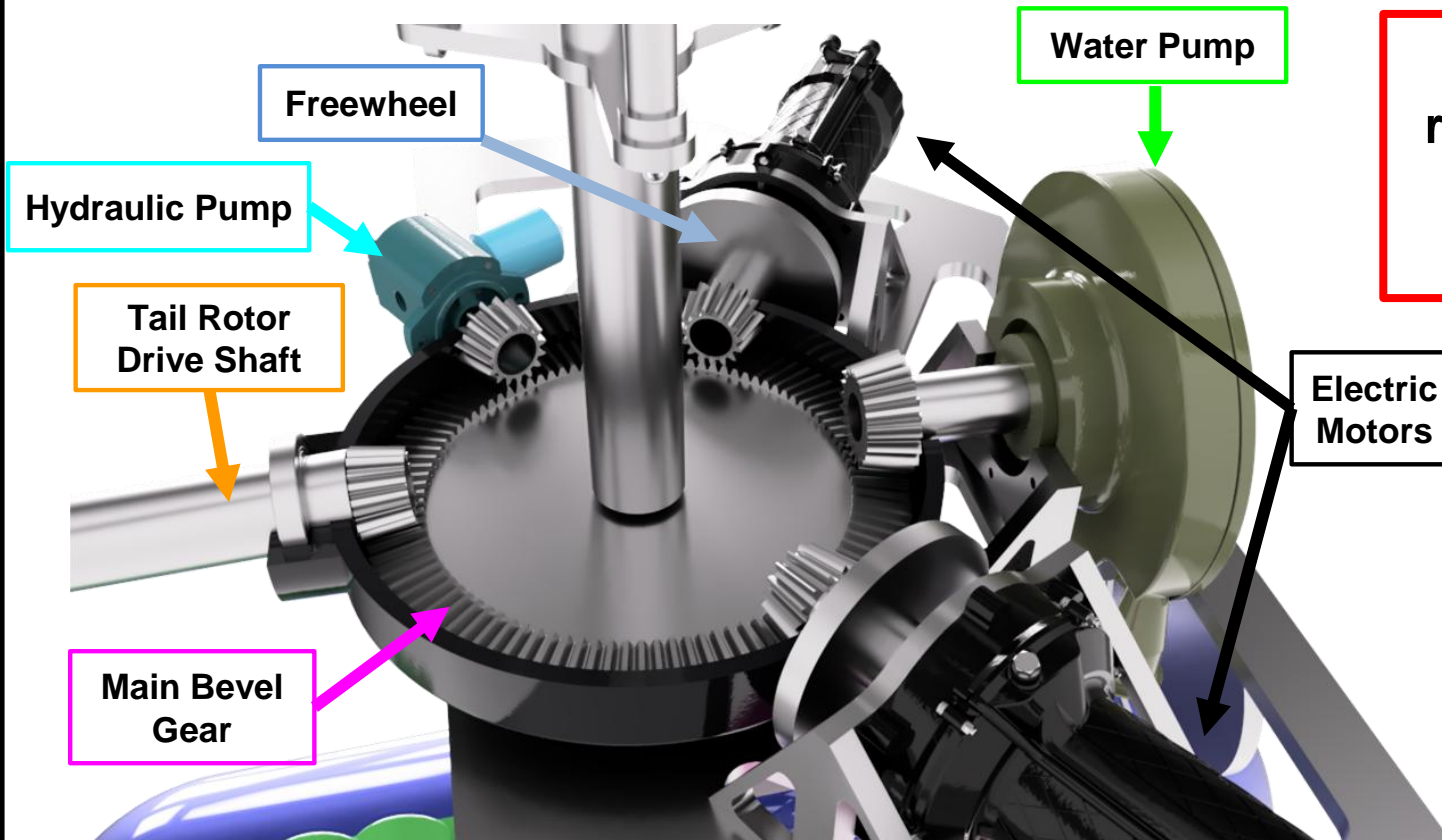
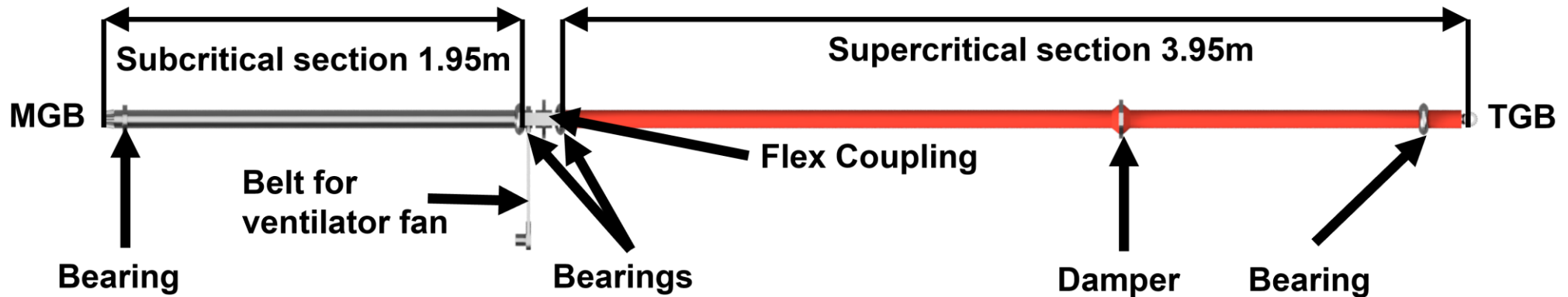
- Battery provides ~25% of power during hover, allowing reduction of PEMFC and cooling system weight
- Powerplant redundancy allows power-assisted autorotation

High Power Flight Battery

- 10C Silicon Anode Cells allow 223 Wh/kg and 2.23 kW/kg on pack level
- 67 kW Nominal, 100 kW Burst
- 20 Ahr capacity
- Pack has two 49S2P modules



Supercritical Electromechanical Drive

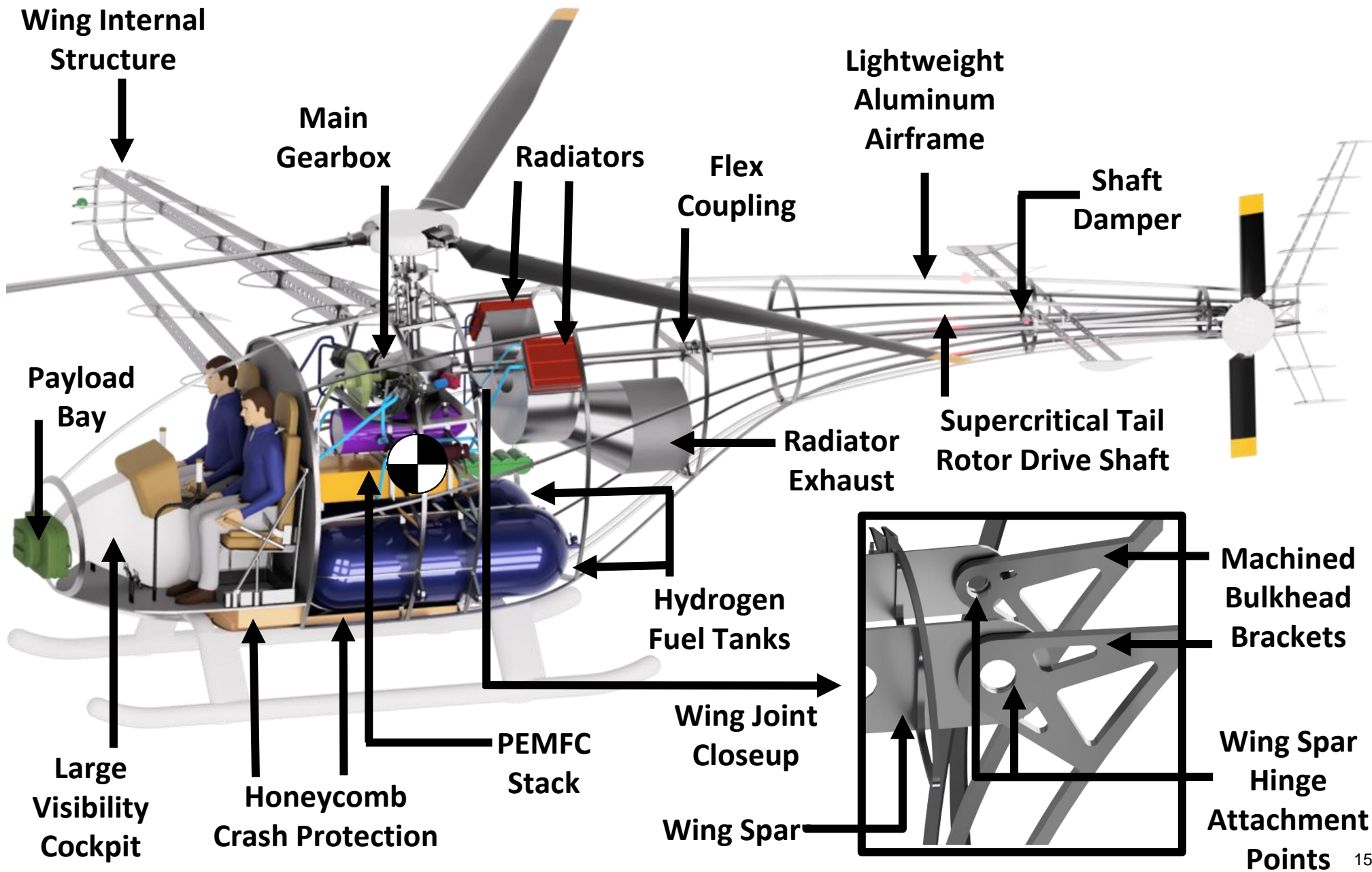


**Supercritical shaft
reduces weight by 50%
compared to a
subcritical shaft**

**Single-stage
gearbox reduces
weight by 70%
compared to a two-
stage reduction**

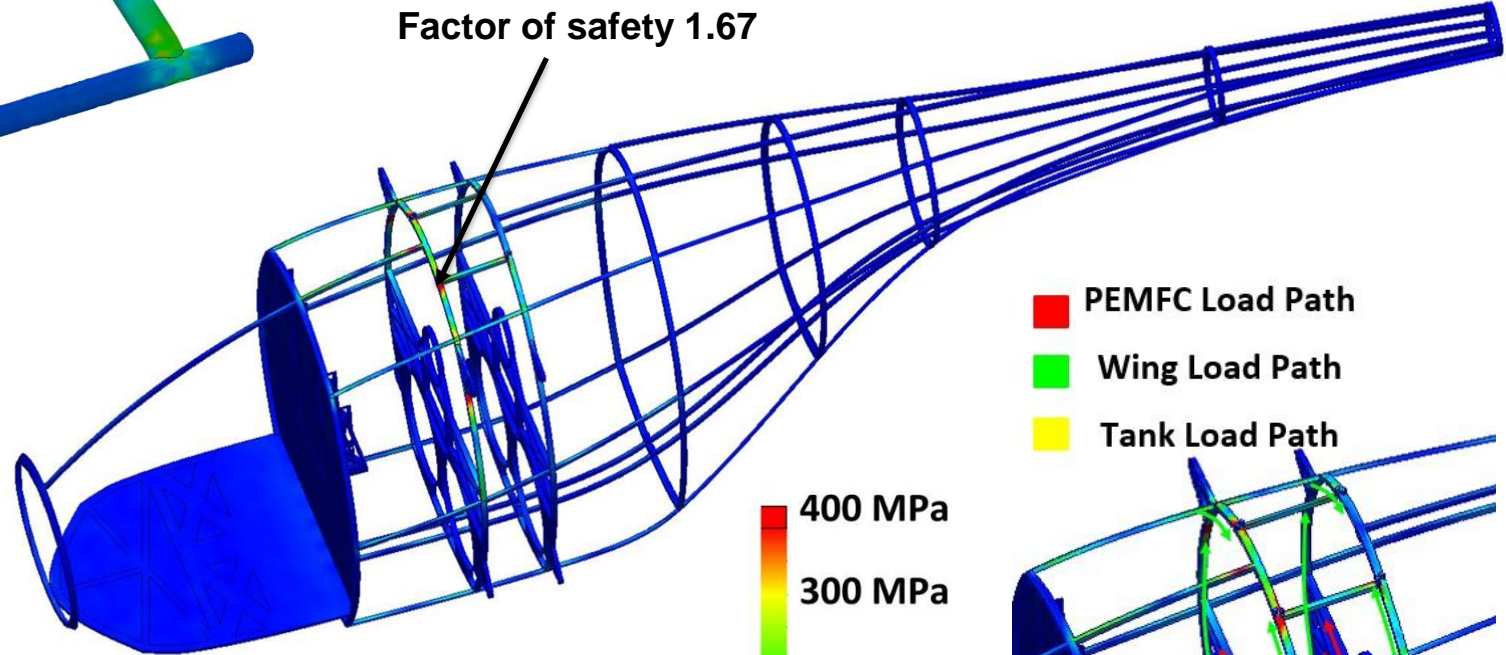
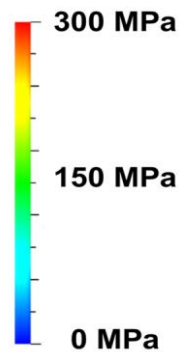
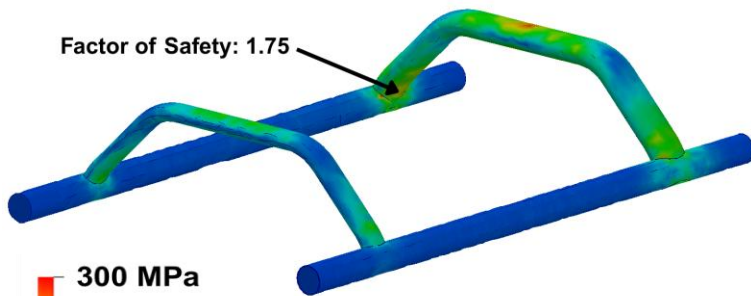


Wyvern Cutaway

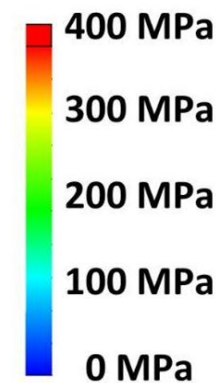


Roll-Over and Resonance-Free Airframe

FEA simulation under 3.5g loads



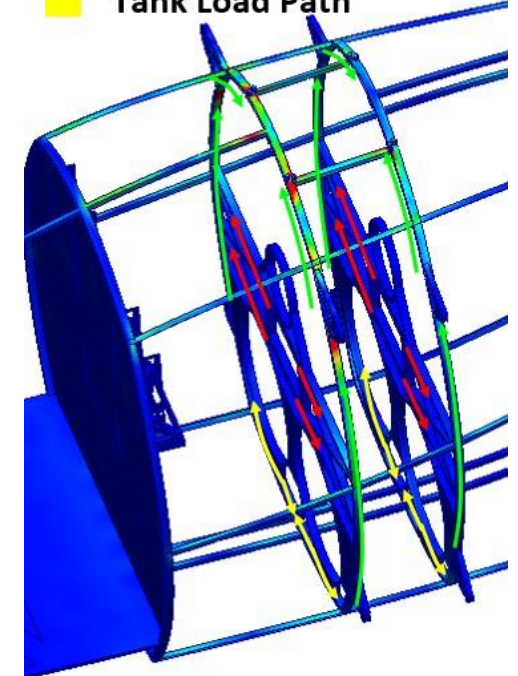
■ PEMFC Load Path
■ Wing Load Path
■ Tank Load Path



40°
Tip-over angle



52° 32°
Pitch angle

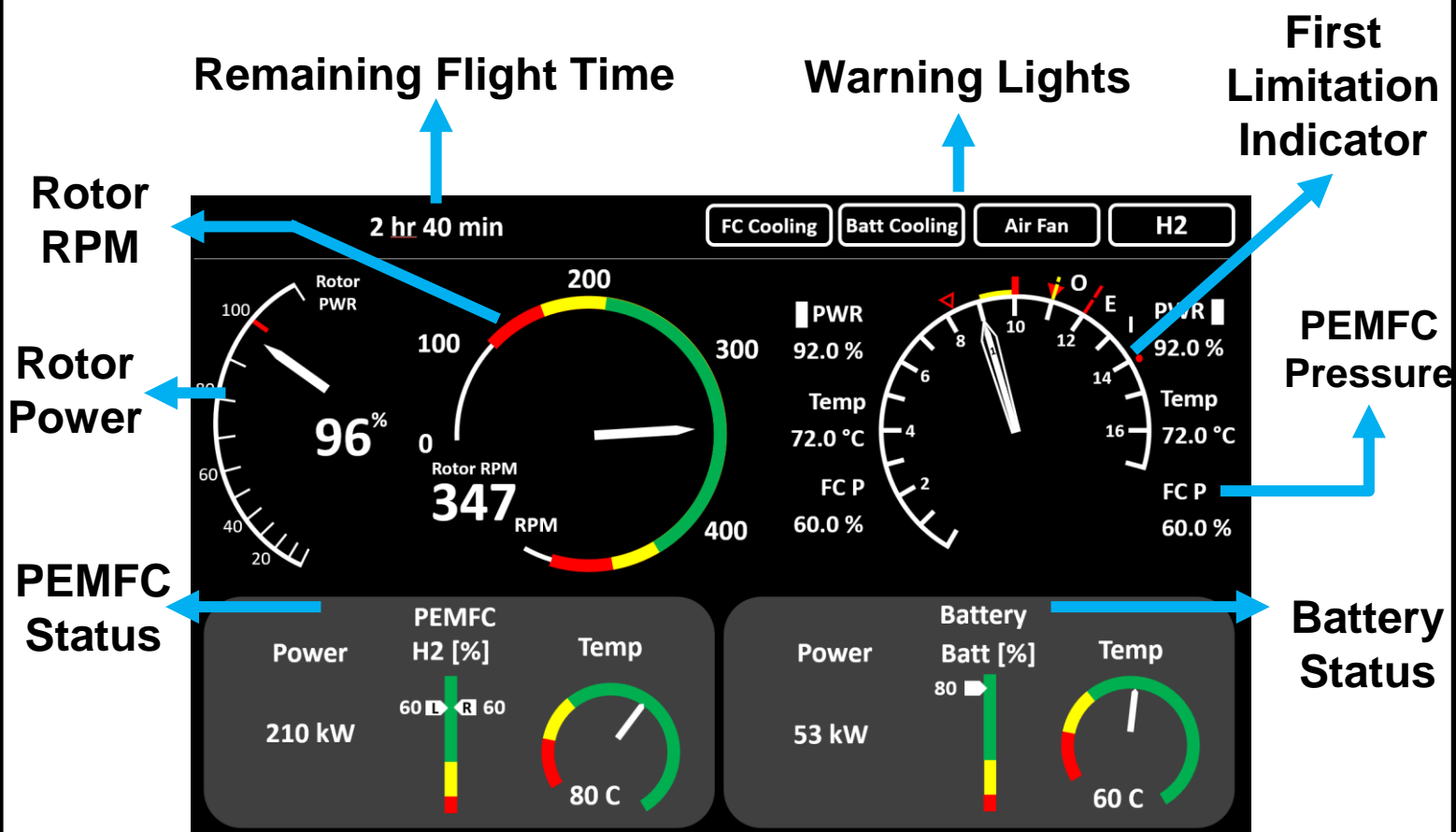




Cockpit Avionics & Advanced Pilot Interface

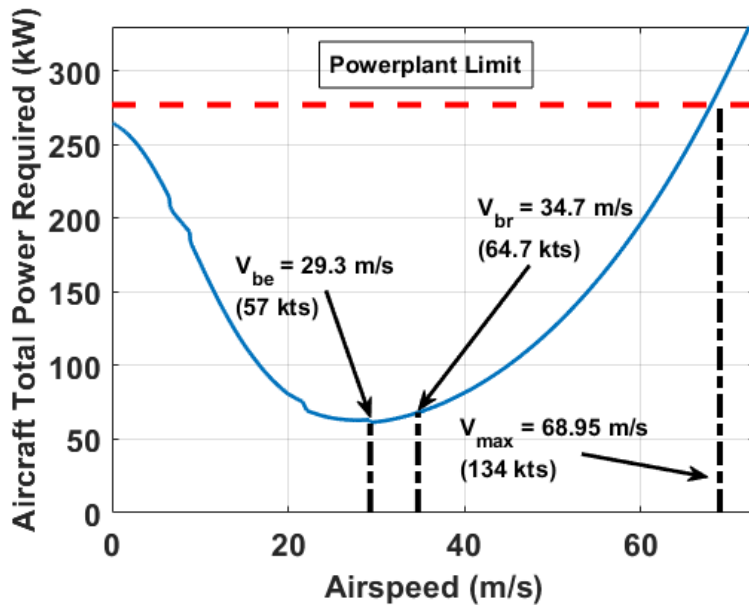


Engine Monitoring System for PEMFC





Unprecedented Loiter, Uncompromised Hover

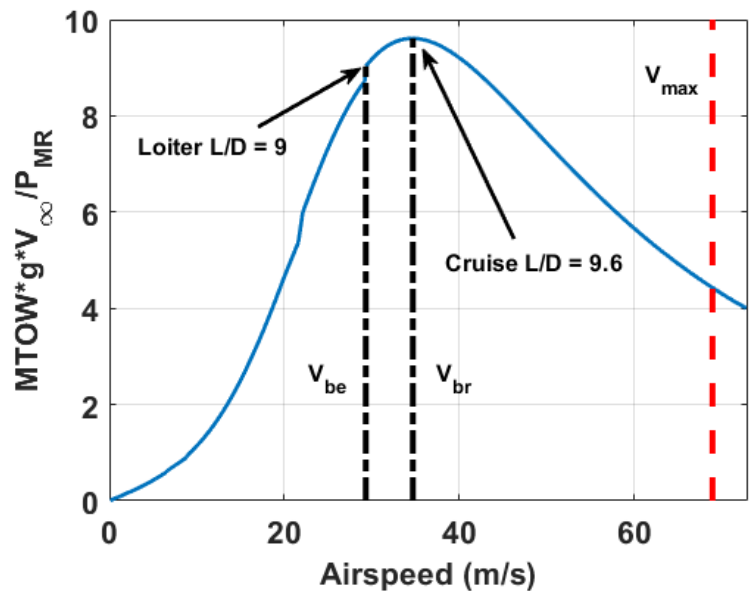


- Highly efficient box wing
- Optimal lift share through slowed rotor
- Streamlined airframe produced:

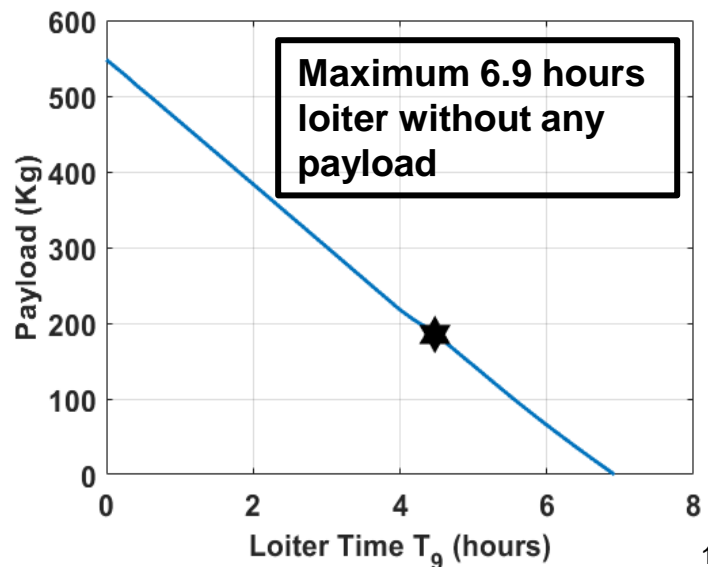
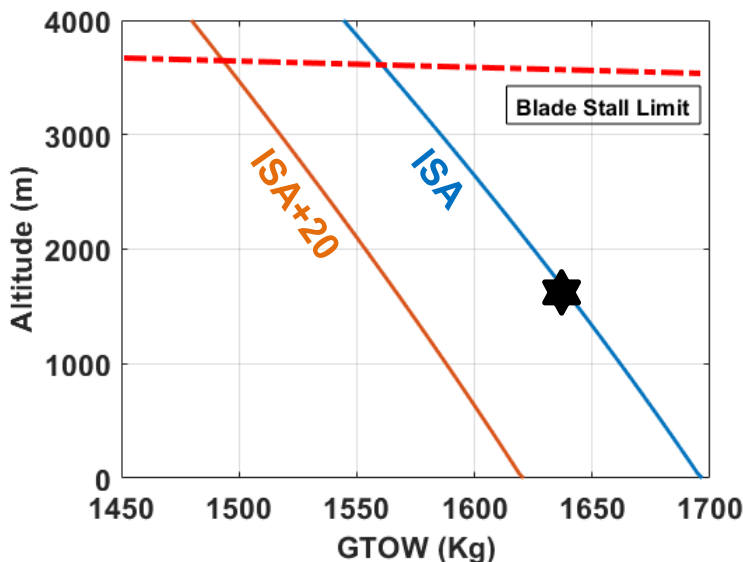
$$(L/D)_e = 9 \text{ in Loiter}$$

Enabling 4.5 hours of Endurance

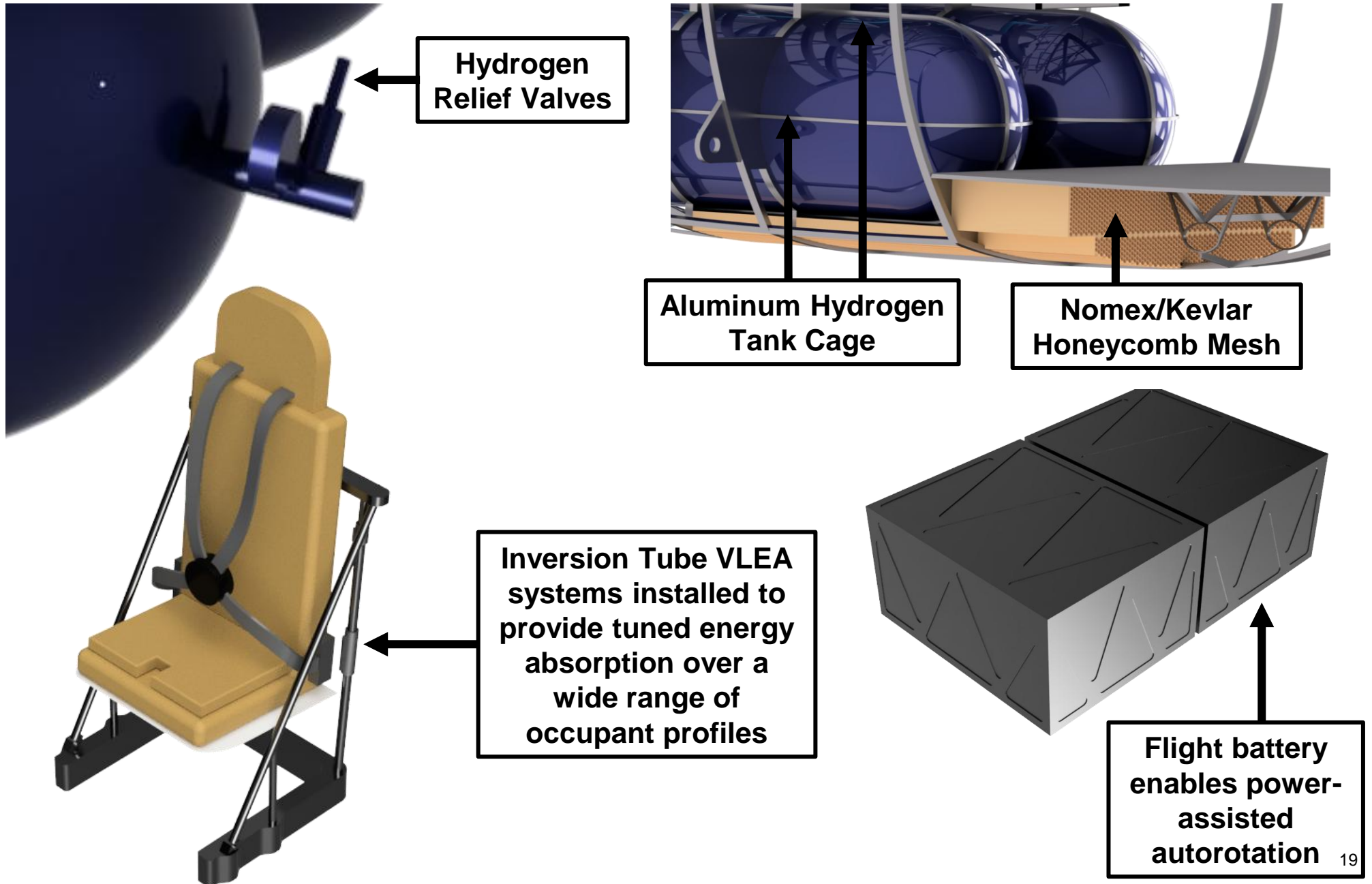
FM prioritized rotor optimization for low specific power stack, ensured exceptional loiter as well as hover performance.



★ HOGE Ceiling at GTOW = 1390 m



Safety Features





Summary

Wyvern is a hydrogen-powered lift-compound SMR designed for long endurance missions.

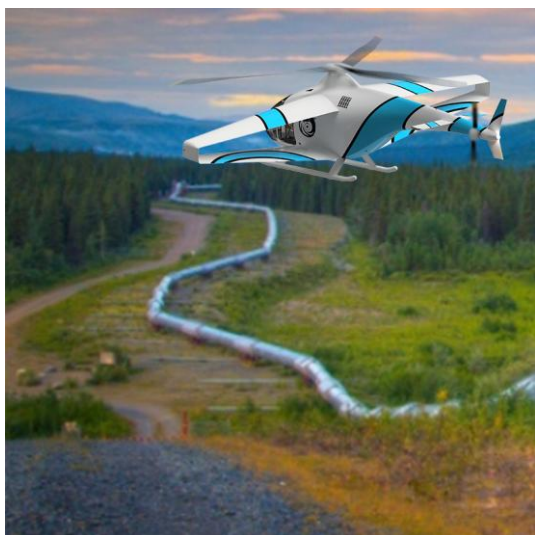
Long endurance capability

- Maximum endurance: 4.5 hours
- Maximum range: 562 km
- Total H₂ required: 24.8kg

Affordability

- 1,193,000 USD
- 734 USD per flight hour

Wyvern's capability for additional missions



Aerial Survey



**Firefighter Command
and Control**



Crop-dusting